



**tyvak**  
Nano-Satellite Systems Inc.

# Interplanetary CubeSat Launch Opportunities and Payload Accommodations

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**Partnered with**  
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**San Luis Obispo, CA**

# Tyvak CubeSat Launch Services

- **Principals Have Extensive Launch Integration Experience**

- **Over a Decade Integrating CubeSats with Launch Vehicles**

- First launch in 2003

- **Successfully Completed 18 Launch Campaigns**

- 121 CubeSat integrated, 72 P-PODs
- 10 Different launch vehicles and 9 ranges worldwide

- **7+ Launch Campaigns Ongoing**

- >40 CubeSats, >25 P-PODs



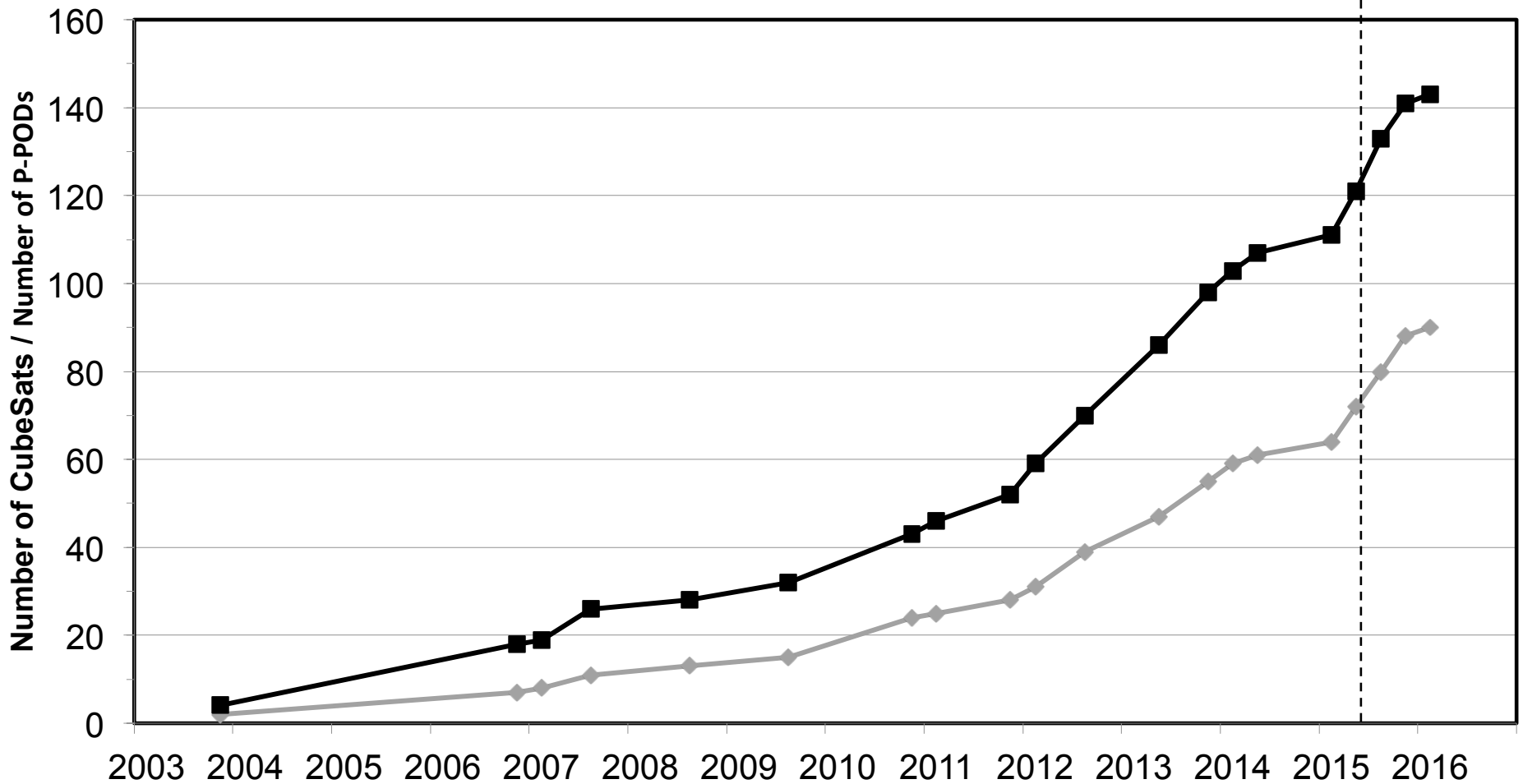
NASA NPP with 3 P-PODs

U.S. Govt with 8 P-PODs



# Extensive POD Flight History

P-PODs Launched: **72** (25+ Manifested)  
CubeSats Launched In CP P-PODs: **121** (40+ Manifested)



# CubeSat Launch Services – Launch Certifications

## • **Launch Coordination and Regulatory Approvals**

- Experienced in ITAR
  - International CubeSats on International LVs, International CubeSats on US LVs
- Radio Frequency Licenses
  - IARU, FCC, NTIA, ITU
- Range Safety AFSPCMAN 91-710
- Flexible Launch Manifesting
- CubeSat On-Orbit Operations

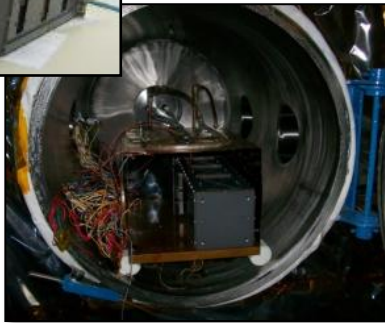
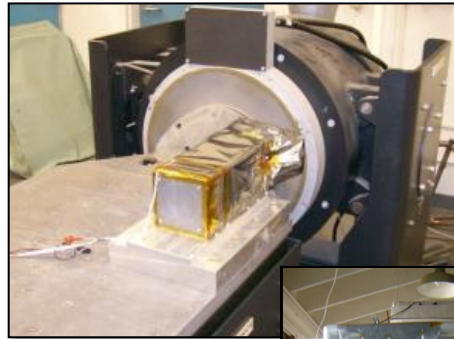
## • **Approvals From U.S./International Government Agencies**

- Successfully managed and assisted CubeSats in obtaining:
  - Export Control: Dept. of State, Commerce
  - CubeSat Frequency Allocations: IARU, FCC, NTIA, ITU
  - Orbital Debris Assessment Report (ODAR) and Re-Entry Analysis: NASA, AF, FCC
  - Earth Imaging: NOAA

# Available Environmental Processing and Testing Facilities

## Vibration Test Facilities

- 2 Vibration Tables
- Used for Qualification and Acceptance Testing of P-POD and CubeSat Hardware
- Random Vibration, Sine Vibe, and Sine Sweeps



## Thermal Vacuum Facilities

- Housed in a Portable Cleanroom
- Temp Range: -60 to 80C
- High Vacuum
- Data Acquisition System
- Used for Qualification, Acceptance, and Thermal Bake Out of Flight Hardware

## Clean Room Facilities

- Class 100,000 Nominal
  - Capable of Class 10,000 and cleaner
- 28 x 8 ft. of Floor Space
- Locked Storage for Flight Hardware

## EMI/EMC Anechoic Chamber Facilities

- Tests for attenuation and CubeSat RF antenna patterns

## Supported CubeSat Testing at Other Facilities

- Commercial Test Facilities, NASA Ames, NASA WFF, NASA MSFC, SRI, NPS

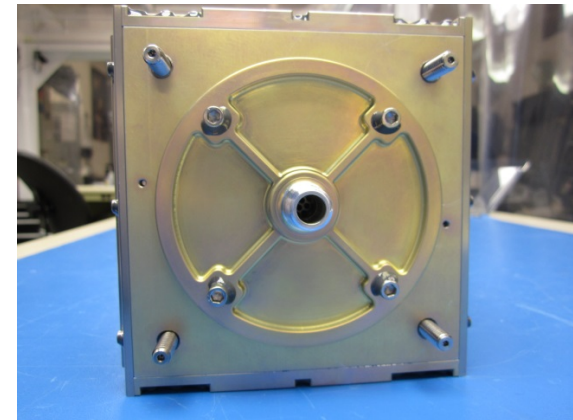
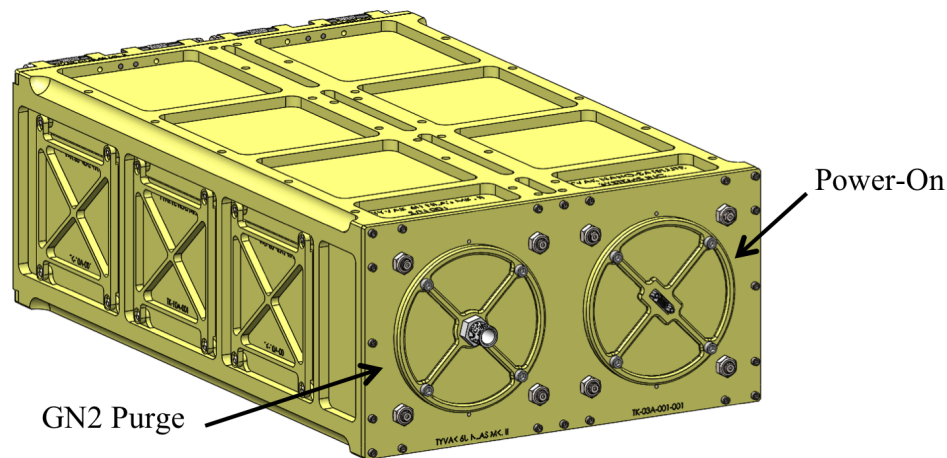
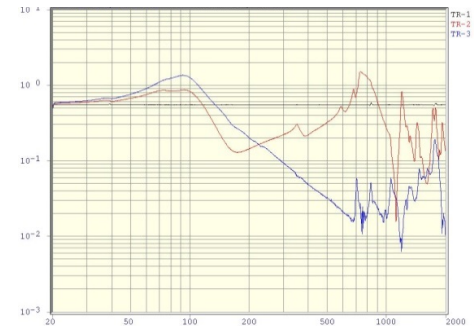
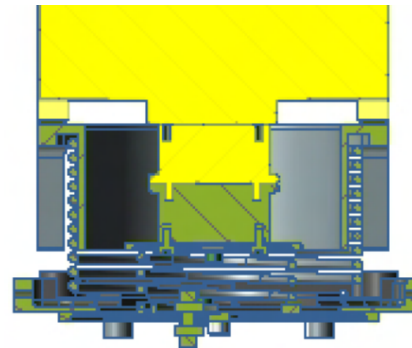


# Possible Capabilities for Interplanetary CubeSats

- **Additional Capabilities for Sensitive CubeSats/Payloads**

- Dependent on LV availability
- Planetary Protection Processing
- Nitrogen Purge
- Trickle Charging and Diagnostics
- Vibration Isolation

- **Looking for feedback from community**



# Current Launch Opportunities

Date	Mission	Destination	Location	Capacity
2015	U.S. Govt	LEO	Atlas V	24U
2015	U.S. Govt	LEO	Atlas V	24U
Q1 2016	Commercial	LEO	Indian	12U
Q1 2016	Commercial	LEO	Russian	12U
Q1 2016	U.S. Govt	Mars	Atlas V	12U
H1 2016	Commercial	LEO	Russian	9U
H1 2016	Commercial	LEO	Indian	12U
H2 2016	Commercial	LEO/Escape	Atlas V	24U+
H2 2016	Commercial	LEO	Indian	12U
H2 2016	Commercial	LEO	Russian	24U+
H1 2017	Commercial	GEO	US	24U+

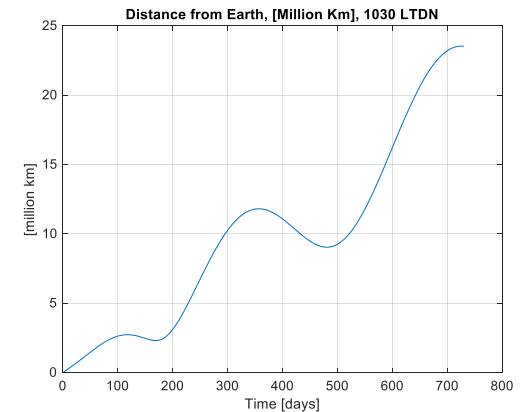
# Current Launch Opportunities - GEO

- **GEO Launch Opportunity**
  - Primary Spacecraft: SS/Loral Satellite Bus
  - Maximum Mass: 60kg
  - Maximum Volume: 1m x 0.5m x 0.4m
  - CubeSat Form Factors: 1U to 27U+
  - Launch: Early Q1 2017
  - Delivery for Integration: ~Sept. 2016
  - Cost: ~\$100,000 USD per kg (negotiable)
  - Contract Signing: June/July 2015



# Current Launch Opportunities - Interplanetary

- **1<sup>st</sup> Commercial Interplanetary Launch Opportunity!**
- **Very Quick Turnaround Needed**
- **Launch Details**
  - Launch Vehicle: ULA Atlas V
  - CubeSat Form Factors: 1U to 12U
  - Capacity Available: 108U
  - Launch: Q3 2016
  - Delivery for Integration: June/July 2016
  - Cost: ~\$700,000 USD per 3U to Elliptical or Escape Orbit
  - Contract Signing: ASAP
- **Orbit Option #1**
  - LEO SSO 620km
- **Orbit Option #2**
  - Highly Elliptical
  - Perigee: ~620km
  - Apogee: Between Lunar and 1,000,000km
  - Phasing possible for lunar flyby
  - 10-40 day orbit period
  - ~40 m/s at perigee to go escape
- **Orbit Option #3**
  - Earth Escape Orbit
  - C3 of ~0.1



# Conclusions

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- **Looking for feedback from Interplanetary CubeSat community on additional requirements and needs**
- **Current interplanetary launch opportunity in Q3 2016, need to act fast if interest/funding is available**
- **Collaboration with science teams creates new opportunities**



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